REDBIRD FLIGHT SIMULATIONS

MX2
QUALIFICATION AND APPROVAL GUIDE (QAG)

ADVANCED AVIATION TRAINING DEVICE

William Jen, Certification Manager

Redbird Flight Simulations, Inc.
2301 East St. Elmo Rd., Suite 100
Austin, Texas 78744

http://www.redbirdflight.com
info@redbirdflight.com

(512) 301-0718 Phone / (512) 301-0770 Fax

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FAA APPROVED QAG
Signature and Date

SHAWN M
HAYES
Digitally signed by
SHAWN M HAYES
Date: 2020.01.16 17:20:09
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Shawn Hayes
Manager, Airmen Training and Certification Branch

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LOG OF REVISIONS

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Original	1.0		CG
Added Log of Revisions	2.0		WJ
Updated Section 1 Device Overview	2.0		WJ
Removed List of Applicable Configurations from Section 1	2.0		WJ
Replaced Section 2 with Overall Aviation Training Device and Description and Pictures	2.0		WJ
Moved Figures subsections from Section 4 to Section 2	2.0		WJ
Replaced Section 3 with Training Device Components Lists	2.0		WJ
Added Win10, Navigator to Table 1 in Section 3	2.0		WJ
Moved Statements of Compatibility to Section 3	2.0		WJ
Replaced Section 4 with ATD Design Criteria	2.0		WJ
Moved Statement of Compliance to Section 4	2.0		WJ
Updated Display Requirements (1) and (2)	2.0		WJ
Added Performance tables to Section 4	2.0		WJ
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Added DA42-G2 configuration for approval	2.0		WJ
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Added Section 6 Visual System details	2.0		WJ
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Consolidated Groups 1 – 5 in Section 5 into Configuration Components	2.0A		WJ
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Updated description and image for Instructor Station	2.1	3,5,6	WJ
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Moved aircrafts submitted in v2.0B into Previously Approved	2.1	15-63	WJ
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Updated Compliance Statement according to new FAA template	2.1A	1	WJ
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Updated/added additional IOS images	2.1A	3	WJ
Added Corvus, TRACON, GIFT to Section 3	2.1A	8	WJ
Deleted Statements of Compatibility of Software and Hardware	2.1A	10	WJ
Updated Section 4 according to new FAA template	2.1A	10-14	WJ
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Updated Performance Tables with V_G according to new FAA template	2.1A	71-72	WJ
Updated Section 7 according to new FAA template	2.1A	75-77	WJ
Corrected configuration typo – Throttle corrected to (T-M)	2.1B	52	WJ
Updated Cover page according to FAA template	2.1C	-	WJ
Updated reference to AC61-136B with AC61-136 per FAA request	2.1C	1, 10, 14	WJ
Corrected typo in B.3.3.1 and B.3.3.2	2.1C	11	WJ
Corrected typo referencing LD, SD, FMX, MCX to MX2	2.1C	14	WJ
Updated Section 7 with the correct text from FAA template	2.1C	77	WJ
Added aircraft configurations to Section 2 according to FAA template	2.1D	2	WJ

SECTION 1: COMPLIANCE STATEMENT

This Qualification and Approval Guide (QAG) provides a detailed description of all the required components, features, functions, and capabilities for the Redbird Flight Simulations, Inc. model MX2 aviation training device. This includes any optional airplane configurations with quality color pictures and diagrams. This QAG is provided by Redbird Flight Simulations, Inc. to clearly describe and verify the required functionality of this aviation training device platform confirming its suitability for airman training and experience. The information as described in advisory circular AC 61-136, FAA Approval of Aviation Training Devices (ATD) and Their Use for Training and Experience is provided within this document. This includes listing all of the required qualifying items, functions, and capabilities. A valid FAA Letter of Authorization (LOA) specifying the credit allowances must accompany the training device when utilized for satisfying airman training or experience requirements specified in 14 CFR §61 or 141. Additionally, FAA Order 8900.1 Volume 11 Chapter 10 Section 1 provides guidance to aviation safety inspectors facilitating ATD evaluations, approvals and oversite.

Redbird Flight Simulations, Inc. will provide a detailed operations manual with each aviation training device model produced. This will include how to properly start, operate, and shut down the trainer. This must include how to operate and maintain the trainer as originally designed and tested. Redbird Flight Simulations, Inc. will ensure that the operator of this training device is familiar and proficient with all the features and capabilities of this trainer, and how to correct any malfunctions that may occur.

The operator of this aviation training device is expected to become proficient in it operation before using it to satisfy any pilot experience requirements specified in the code of federal regulations. This includes maintaining its condition and functionally. This ATD must be maintained to its original performance and functionality, as demonstrated during the original FAA functional evaluation. This trainer cannot be used to log pilot time unless all the components of the trainer are in normal working order.

Only the airplane configurations approved for this model can be utilized when satisfying FAA experience or training requirements. Any additions, changes, or modifications to this model, or the associated configurations, must be evaluated and approved in writing by the General Aviation and Commercial Division. This does not prohibit software updates that do not otherwise change the appearance of the systems operation. Operators who use these trainers to satisfy FAA pilot training or experience requirements specified in part 61 or 141 are obligated to allow FAA inspection ensuring acceptable function and compliance.

Any questions concerning FAA approval or use of ATDs should be directed to the General Aviation and Commercial Division.

SECTION 2: AVIATION TRAINING DEVICE (ATD) DESCRIPTION AND PICTURES

The Redbird MX2 model is based on the dimensions and layout of a several production Single and Multi-Engine Land aircrafts. These models closely represent the overall functionality, performance, avionics, and instrumentation. The platform consists of a cockpit section, instructor's control station, visual display system and an audio system. It incorporates a combination of hardware and software components that is assembled and checked by Redbird Flight Simulations. All hardware elements are permanently installed and designed so the cockpit has the appearance and feel of an actual aircraft. From the pilot's seated position, there are no computer hardware elements such as keyboards, pointing devices, etc. for his or her use.

Redbird MX2 model provides a realistic flight deck design, avionics interface, and reliable hardware/software performance. This platform provides an effective training environment for students and pilots in training. This includes the ability to accomplish scenario based flight training activities, instrument procedures and experience, pilot proficiency evaluations, simulated equipment failure, emergency procedures, and facilitates increased pilot competency.

Airplane Single Engine Land representing:

Beechcraft (A36) Bonanza	Cessna 206 Stationair	Diamond DA40	Piper PA-28-181 Archer III
Cessna 172 Skyhawk	Cessna 206 Stationair TC	Liberty XL2	Piper PA-28R-201 Arrow
Cessna 172RG Cutlass RG	Cirrus SR20	Mooney M20TN (Type S)	Piper PA-38-112 Tomahawk
Cessna 182 Skylane	Cirrus SR22	Maul MX7-180B	Socata TB10
Cessna 182RG Skylane RG	Diamond DA20	Piper PA-28-161 Warrior III	Tecnam P2010

Multi-Engine Land representing:

Beechcraft (BE55) Baron	Diamond DA42 L360	Piper PA-34T Seneca	Tecnam P2006
Beechcraft (BE58) Baron	Diamond DA42 NG	Piper PA-44 Seminole	
Beechcraft (BE76) Duchess	Piper PA-34 Seneca	Piper PA-44T Seminole	

The Redbird MX2 is a versatile, expandable and affordable device that has been designed to be representative of popular single and twin-engine piston aircraft. They offer the user the ability to change the cockpit controls and aircraft models to represent a wide range of popular training aircraft. The line is equipped with the following notable features:

- Single pilot controls with optional 2-axis control-loaded yoke and interconnected rudder pedals
- Enclosed cockpit with pilot and copilot seating
- 3-axis electric motion platform providing pitch, roll and yaw motions
- Wrap-around exterior visuals provided by 5 LCD screens
- Realistic switches, buttons, knobs, circuit breakers and other cockpit controls that are designed to represent the family of aircraft.
- An interchangeable instrument panel, throttles, and yokes to allow a quick-change configuration options to the end user.
- Closed Circuit intercom system, allowing for communication between the pilot, co-pilot and instructors using standard aviation headsets
- A portable instructors station, allowing the instructor to operate from inside or outside the simulator

Configuration Components

Instructor's Station

The Redbird Instructor Station interface is operated through any PC or browser enabled device.
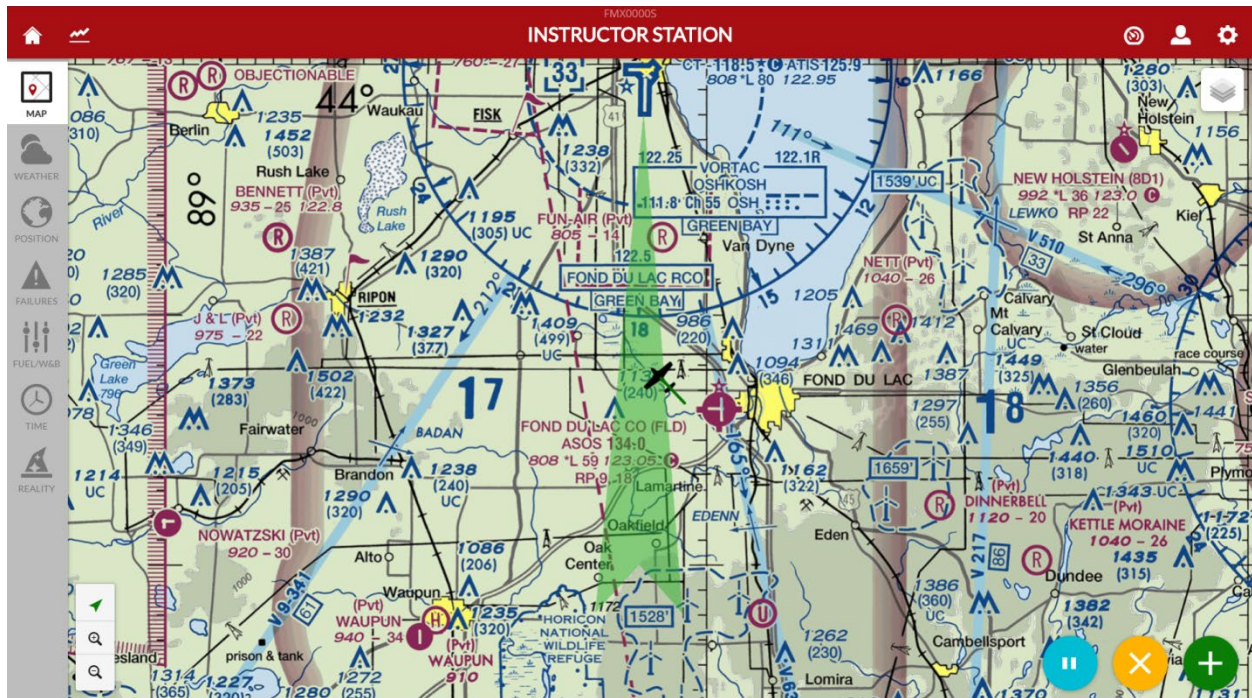

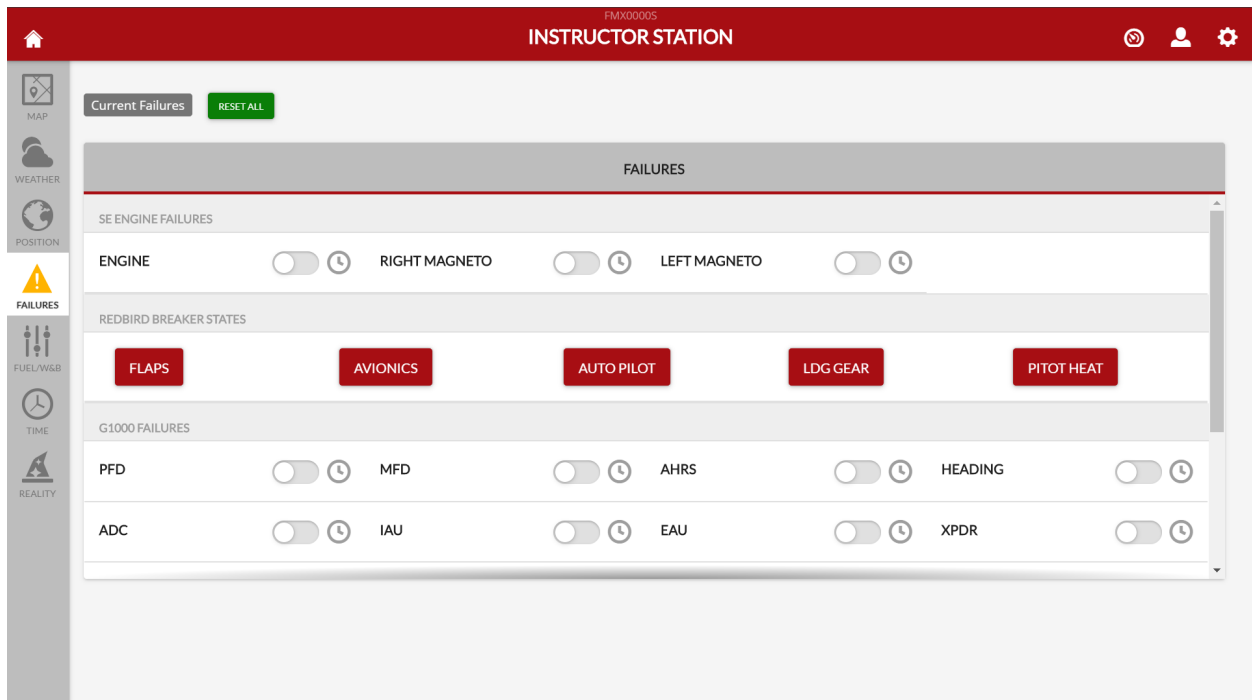Image 5.2: Redbird's Instructor's interface Map Tab

Image 5.2A: Redbird's Instructor's interface Failure Tab

Aircraft Instrument Configurations

The Redbird MX2 simulator is capable of supporting multiple aircraft configurations utilizing both traditional analog instruments as well as glass panels.

Image 5.12: Traditional 6-pack Instrument Panel Configuration (Example)

Image 5.13: Glass Panel Instrument Configuration (Example)

Modular Controls

The modular design of the Redbird MX2 simulator allow for quick configuration changes between single and multi-engine aircraft configurations.
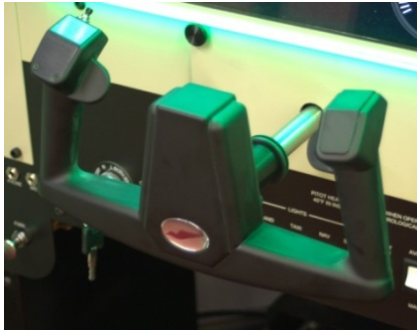
Image 2.1: Center Yoke

Image 2.2: Side Yoke

Image 2.3: Center Stick

Image 3.1: Single Engine Vernier Throttle Quadrant

Image 3.2: Single Engine Complex Vernier Throttle Quadrant

Image 3.3: Single Engine "Boat Style" Throttle Quadrant
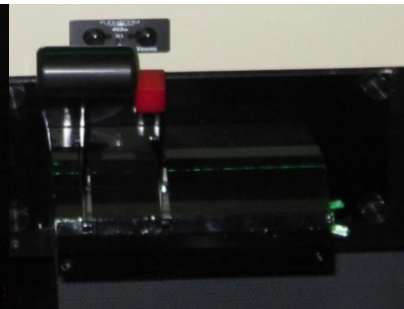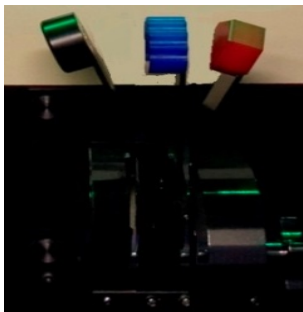
Image 3.4: Single Engine Complex "Boat Style" Throttle Quadrant
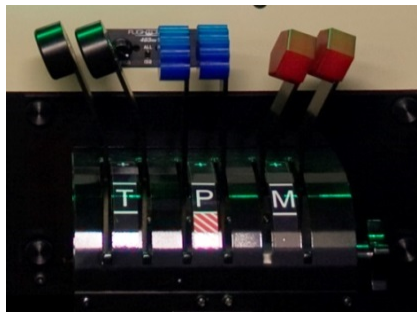
Image 3.5: Twin Engine Throttle Quadrant – Type 1
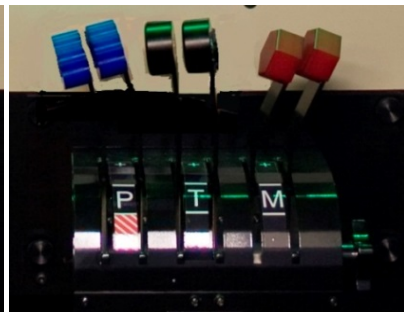
Image 3.6: Twin Engine Throttle Quadrant – Type 2
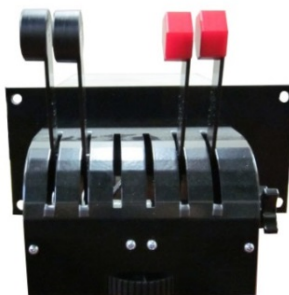
Image 3.7: Twin Engine Throttle Quadrant – Type 3
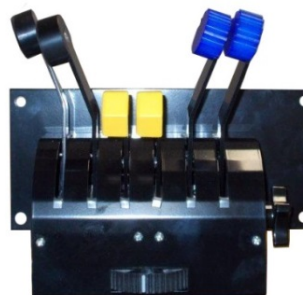
Image 3.8: Twin Engine Throttle Quadrant – Type 4

Image 6.1: Optional CAPS for Cirrus Configurations

Image 6.2: Optional CAPS for Cirrus Configurations

Image 4.1: SwitchPanel (Lower Left)

Image 4.3: Manual Trim Wheel

Image 4.2: SwitchPanel (Lower Right)

Image 4.7: Starter Switch*

Image 4.8: Starter Switch*

Image 5.1: Training Session Buttons*

Image 5.5: Training Session Buttons*

Image 5.8: Dual Starter Retrofit Kit

*May vary by serial number

Image 5.9: Fuel Selector*

Image 5.10: Fuel Selector*

Image 5.3: Rudder Pedals

*May vary by serial number

Image 5.6: Example Assembled Cockpit Configuration

Image 5.7: External Shroud and Motion Platform Type 2

SECTION 3: TRAINING DEVICE COMPONENT LIST

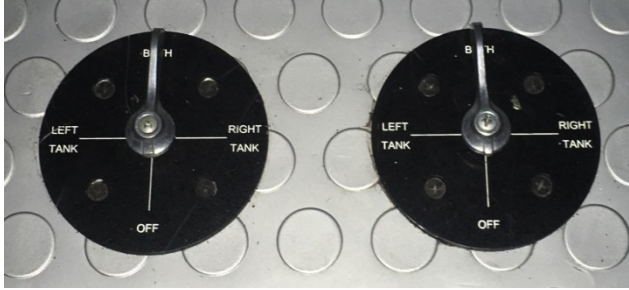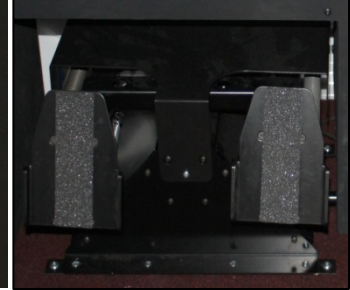| Software Component List					
Qty	**Type**	**Manufacturer**	**Name**	**Description/Function**	**Configuration**
1	Software	Microsoft	Windows XP, 7, or 10	Operating system. (depending on serial number)	All
1	Software	Microsoft	ESP or FSX	Simulation engine. (depending on serial number)	All
1	Software	Lockheed Martin	Prepar3D	Simulation engine. (depending on serial number)	All
1	Software	Redbird Flight Simulations	RB Sim	Simulation control and component integration.	All
1	Software	Redbird Flight Simulations	MX2 Control	Motion system control and component integration.	All
1	Software	Redbird Flight Simulations	Instructors Station	Environmental, location and failure controls with map, track and glideslope display.	All
1	Software	Redbird Flight Simulations	Navigator	Simulation control and component integration. Environmental, location and failure controls with map, track and glideslope display.	All
1	Software	Redbird Flight Simulations	TRACON	Optional ATD Network Management Console	All
1	Software	Redbird Flight Simulations	Parrot	Optional software providing ATC services	All
1	Software	Redbird Flight Simulations	Cygnus (including Pro)	Optional software providing location services	All
1	Software	Redbird Flight Simulations	Corvus	Optional software providing location services	All
1	Software	Redbird Flight Simulations	GIFT	Optional software providing guided flight training	All
1	Software	RealNav Data	Instrument Procedures Database	Provides for FAA published instrument navigation procedures, database per 14 CFR 97 (enroute, approach)	All
1	Software	Mindstar Aviation	Redbird 1000	Virtual replication of the Garmin G1000 flight instruments, GPS, radios, gauges, indicators, alerts, misc. instruments and logic controls for simulated systems.	See Configuration
1	Software	Mindstar Aviation	Redbird 430/ 530	Virtual replication of the Garmin GNS430 and GNS530	See Configuration
1	Software	Mindstar Aviation	Redbird KLN94	Virtual replication of the BendixKing KLN94	See Configuration
1	Software	Mindstar Aviation	Redbird EFD1000	Virtual replication of the Aspen EFD1000	See Configuration
1	Software	Mindstar Aviation	Redbird Entegra	Virtual replication of the Avidyne Entegra flight instruments, radios, gauges, indicators, alerts, misc. instruments and logic controls for simulated systems.	See Configuration
1	Software	Mindstar Aviation	Redbird Autopilot	Virtual replication of the BendixKing KAP140, Genesys STEC55x, and Garmin GFC700 (Redbird 140, 55, or 700)	See Configuration
1	Software	Mindstar Aviation	Redbird Radios	Virtual Radios replication of the BendixKing KX55 and Garmin SL40 (KX155, SL40)	See Configuration
1	Software	Mindstar Aviation	Miscellaneous Gauges	Virtual Miscellaneous Gauges	See Configuration
1	Software	Flight 1	Compass	Virtual Generic Compass	All
1	Software	Redbird Flight Simulations	Analog Gauges	Virtual Airspeed, Attitude, Altimeter, Turn and Bank, Heading, HSI, VSI, RMI, CDI, and ADF Gauges , Radios	See Configuration
1	Software	Redbird Flight Simulations	Miscellaneous Gauges	Virtual Miscellaneous Gauges	All

Table 1: Training Device Component List (Software)

Hardware Component List					
Qty	Type	Manufacturer	Name	Description/Function	Configuration
1	Hardware	Redbird Flight Simulations	Simulation Computer	Host computer for flight simulation engine, simulation control software, airplane systems and instruments.	All
1	Hardware	Redbird Flight Simulations	Motion Control Computer	Host computer for MX2 Control.	All
1	Hardware	Industry Standard	Instructors Station Computer or Wireless Mobile Device	Host computer for Instructors Station.	All
1	Hardware	Redbird Flight Simulations	Motion Platform: Type 2	Gimbaled, steel motion platform with movement in pitch, roll and yaw. Includes all motors, sensors and safety controls.	All
1	Hardware	Redbird Flight Simulations	Cockpit Enclosure	Cockpit enclosure to exclude distractions.	All
5	Hardware	22" LCD Industry Standard Monitor	Visual Display 1	Flat Panel displays for exterior views.	All – Varies by configuration
2	Hardware	19" LCD Industry Standard Monitor	LCD	Flat Panel displays for virtual instruments.	All
1	Hardware	Redbird Flight Simulations	Yoke, Side Yoke, Center Stick	Pitch and roll controller with switches and buttons for airplane systems operation. As required for each configuration.	All – Varies by configuration
1	Hardware	Redbird Flight Simulations	Yoke, Side Yoke, Center Stick (Single) – Control Loading	Optional Control loading pilot yoke with pitch and roll controller and switches and buttons for airplane systems operation. As required for each configuration.	All – Varies by configuration
1	Hardware	Redbird Flight Simulations	Throttle Quadrant	As required for each configuration.	All- See Configuration
1	Hardware	Redbird Flight Simulations	Rudder Pedals - Single	Rudder control pedals with toe brakes.	All – Varies by configuration
1	Hardware	Redbird Flight Simulations	Rudder Pedals (Single) – Control Loading	Optional Control loading pilot rudder control pedals with toe brakes.	All – Varies by configuration
1	Hardware	Redbird Flight Simulations	Switch Panel: MX2	Lower switch panel with airplane configuration and systems controls.	All
1	Hardware	Redbird Flight Simulations	Instrument Controls Overlay	Flight instruments, radios, airplane configuration and systems controls as required for each configuration.	All- See Configuration
1	Hardware	Redbird Flight Simulations	Dual Starter Retrofit Kit	Dual Starter, battery, and ALT switches for Switch Panel 1	Optional - All
1	Hardware	Redbird Flight Simulations	CAPS	Optional CAPS	All – See Configuration

Table 2: Training Device Component List (Hardware)

SECTION 4: AVIATION TRAINING DEVICE (ATD) DESIGN CRITERIA LIST

The following section provides the detailed "word for word" listing and design criteria of each of the required items, functions, and capabilities (listed in AC 61-136, for BATD requirements Appendix B and the additional AATD items of Appendix C) and operational performance value/scale (as applicable) for each of the functions described for the Redbird MX2.

Basic ATD Requirements List [Appendix 2 items]

All configurations for this model, as noted, meet AC 61-136, Appendix B requirements

The Redbird MX2 meets the following Control Input Requirements:

B.3.1.1 The aircraft physical flight and associated control systems ARE recognizable as to their function and how they are to be manipulated solely from their appearance. These physical flight control systems DO NOT use interfaces such as a keyboard, mouse, or gaming joystick to control the aircraft in simulated flight.

B.3.1.2 Virtual controls are those controls used to set up certain aspects of the simulation (such as selecting the aircraft configuration, location, weather conditions, etc.) and otherwise program, effect, or pause the training device. These controls ARE part of the instructor station or independent computer interface.

B.3.1.3 Except for the initial setup, a keyboard or mouse IS not be used to set or position any feature of the ATD flight controls for the maneuvers or training tasks to be accomplished. See the control requirements listed below as applicable to the aircraft model represented. The pilot IS able to operate the controls in the same manner as it would be in the actual aircraft. This includes the landing gear, wing flaps, cowl flaps, carburetor heat, mixture, propeller, and throttle controls appropriate to the aircraft model represented.

B.3.1.4 The physical arrangement, appearance, and operation of controls, instruments, and switches closely MODELS the aircraft represented. The Redbird MX2 recreates the appearance, arrangement, operation, and function of realistically placed physical switches and other required controls representative of an aircraft instrument panel that includes the following:

- Master/battery;
- Magnetos for each engine (as applicable);
- Alternators or generators for each engine;
- Auxiliary power unit (APU) (if applicable);
- Fuel boost pumps/prime boost pumps for each engine;
- Avionics master;
- Pitot heat; and
- Rotating beacon/strobe, navigation, taxi, and landing lights.

B.3.1.5 ONLY the software evaluated by the FAA may be loaded for use on that computer system. This does not PROHIBIT software updates that do not otherwise change the appearance of the systems operation.

The Redbird MX2 meets the following additional airplane physical flight and airplane systems controls:

B.3.2.1.1 A **self-centering displacement yoke or control stick** that allows continuous adjustment of pitch and bank.

B.3.2.1.2 **Self-centering rudder pedals** that allow continuous adjustment of yaw and corresponding reaction in heading and roll.

B.3.2.1.3 **Throttle or power control(s)** that allows continuous movement from idle to full-power settings and corresponding changes in pitch and yaw, as applicable.

B.3.2.1.4 **Mixture/condition, propeller, and throttle/power control(s)** as applicable to the M/M of aircraft represented.

B.3.2.1.5 Controls for the following items, as applicable to the category and class of aircraft represented:

- Wing flaps,
- Pitch trim,

- Communication and navigation radios,
- Clock or timer,
- Gear handle (if applicable),
- Transponder,
- Altimeter,
- Carburetor heat (if applicable), and
- Cowl flaps (if applicable).

The Redbird MX2 meets the following Control Input Functionality and Response Criteria:

B.3.3.1 Time from control input to recognizable system response IS without delay AND DOES NOT not appear to lag in any way. Redbird Flight Simulations, Inc. verifies that the Redbird MX2 meets this requirement.

B.3.3.2 The control inputs ARE tested by the computer and software program at each startup and displayed as a confirmation message of normal operation or a warning message IF the transport delay time or any design parameter is out of tolerance. It IS not possible to continue the training session unless the problem is resolved and all components are functioning properly. This test considers all the items listed in the display and control requirements.

The Redbird MX2 meets the following Display Requirements:

B.3.4.1 The following instruments and indicators ARE replicated and properly located as appropriate to the aircraft represented:

B.3.4.1.1 Flight instruments ARE in a standard configuration representing the traditional "round" dial flight instruments or as an electronic primary flight instrument display (PFD) and multi-function display (MFD) with reversionary and back-up flight instruments.

B.3.4.1.2 A sensitive **altimeter** with incremental markings each 20 feet or less, operable throughout the normal operating range of the M/M of aircraft represented.

B.3.4.1.3 A **magnetic direction indicator**.

B.3.4.1.4 A **heading indicator** with incremental markings each 5 degrees or less, displayed on a 360 degree circle. Arc segments of less than 360 degrees ARE selectively displayed as applicable to the M/M of aircraft represented.

B.3.4.1.5 An **airspeed indicator** with incremental markings as shown for the M/M aircraft represented; airspeed markings of less than 20 knots need not be displayed.

B.3.4.1.6 A **vertical speed indicator** (VSI) with incremental markings each 100 feet per minute (fpm) for both climb and descent, for the first 1,000 fpm of climb and descent, and at each 500 fpm climb and descent for the remainder of a minimum ±2,000 fpm total display, or as applicable to the M/M of aircraft being represented.

B.3.4.1.7 A **gyroscopic rate-of-turn indicator** or equivalent with appropriate markings for a rate of 3 degrees per second turn for left and right turns. If a turn and bank indicator is used, the 3 degrees per second rate index IS inside of the maximum deflection of the indicator.

B.3.4.1.8 A **slip and skid indicator** with coordination information displayed in the conventional inclinometer format where a coordinated flight condition is indicated with the ball in the center position. A split image triangle indication as appropriate for PFD configurations may be used.

B.3.4.1.9 An **attitude indicator** with incremental markings each 5 degrees of pitch or less, from 20 degree pitch up to 40 degree pitch down or as applicable to M/M of aircraft represented. Bank angles ARE identified at "wings level" and at 10, 20, 30, and 60 degrees of bank (with an optional additional identification at 45 degrees) in left and right banks.

B.3.4.1.10 **Engine instruments** as applicable to the M/M of aircraft being represented, providing markings for the normal ranges including the minimum and maximum limits.

B.3.4.1.11 A **suction gauge** or instrument pressure gauge with a display applicable to the aircraft represented.

B.3.4.1.12 A **flap setting indicator** that displays the current flap setting. Setting indications should be typical of that found in an actual aircraft.

B.3.4.1.13 A **pitch trim indicator** with a display that shows zero trim and appropriate indices of airplane nose down and airplane nose up trim, as would be found in an aircraft.

B.3.4.1.14 **Communication radio(s)** with a full range of selectable frequencies displaying the radio frequency in use.

B.3.4.1.15 **Navigation radio(s)** with a full range of selectable frequencies displaying the frequency in use and capable of replicating both precision and nonprecision instruments, including approach procedures (each with an aural identification feature), and a marker beacon receiver. For example, an instrument landing system (ILS), non-directional radio beacon (NDB), Global Positioning System (GPS), Localizer (LOC) or very high frequency omni-directional range (VOR). Graduated markings as indicated below ARE present on each course deviation indicator (CDI) as applicable. The marking include:

- One-half dot or less for course/glideslope (GS) deviation (i.e., VOR, LOC, or ILS), and
- Five degrees or less for bearing deviation for automatic direction finder (ADF) and radio magnetic indicator (RMI), as applicable.

B.3.4.1.16 A clock with incremental markings for each minute and second, or a timer with a display of minutes and seconds.

B.3.4.1.17 A transponder that displays the current transponder code.

B.3.4.1.18 A fuel quantity indicator(s) that displays the fuel remaining, either in analog or digital format, appropriate for M/M of aircraft represented.

B.3.4.2 All instrument displays listed above ARE visible during all flight operations. Allowances can be made for multifunction electronic displays that may not display all instruments simultaneously. All of the displays must provide an image of the instrument that is clear and:

B.3.4.2.1 Does not appear to be out of focus or illegible.

B.3.4.2.2 Does not appear to "jump" or "step" during operation.

B.3.4.2.3 Does not appear with distracting jagged lines or edges.

B.3.4.2.4 Does not appear to lag relative to the action and use of the flight controls.

B.3.4.3 Control inputs ARE PROPERLY reflected by the flight instruments in real time and without a perceived delay in action. Display updates must show all changes (within the total range of the replicated instrument) that are equal to or greater than the values stated below:

B.3.4.3.1 Airspeed indicator: change of 5 knots.

B.3.4.3.2 Attitude indicator: change of 2 degrees in pitch and bank.

B.3.4.3.3 Altimeter: change of 10 feet.

B.3.4.3.4 Turn and bank: change of ¼ standard rate turn.

B.3.4.3.5 Heading indicator: change of 2 degrees.

B.3.4.3.6 VSI: change of 100 fpm.

B.3.4.3.7 Tachometer: change of 25 rpm or 2 percent of turbine speed.

B.3.4.3.8 VOR/ILS: change of 1 degree for VOR or ¼ of 1 degree for ILS.

B.3.4.3.9 ADF: change of 2 degrees.

B.3.4.3.10 GPS: change as appropriate for the model of GPS-based navigator represented.

B.3.4.3.11 Clock or timer: change of 1 second.

B.3.4.4 Displays must reflect the dynamic behavior of an actual aircraft (e.g., a VSI reading of 500 fpm must reflect a corresponding movement in altitude, and an increase in power must reflect an increase in the rpm indication or power indicator.)

The Redbird MX2 meets the following Flight Dynamics Requirements:

B.3.5.1   Flight dynamics of the ATD ARE comparable to the way the represented training aircraft performs and handles. However, there is no requirement for an ATD to have control loading to exactly replicate any particular aircraft.

B.3.5.2   Aircraft performance parameters (such as maximum speed, cruise speed, stall speed, maximum climb rate, and hovering/sideward/forward/rearward flight) ARE comparable to the aircraft being represented. A performance table IS included in the QAG for each aircraft configuration for sea level and 5,000 feet using standard atmosphere and gross weight conditions, to verify the appropriate performance. An alternate performance altitude for 6,000 feet can be used if the manufacturer of that aircraft has a performance chart reflecting that altitude. Performance at altitude for turboprop or turbojet configurations should reflect 18,000 ft.

B.3.5.3   Aircraft vertical lift component CHANGES as a function of bank comparable to the way the aircraft being represented performs and handles.

B.3.5.4   Changes in flap setting, slat setting, gear position, collective control, or cyclic control ARE accompanied by changes in flight dynamics comparable to the way the M/M of aircraft represented performs and handles.

B.3.5.5   The presence and intensity of wind and turbulence ARE reflected in the handling and performance qualities of the simulated aircraft and IS comparable to the way the aircraft represented performs and handles.

The Redbird MX2 meets the following Instructional Management Requirements:

B.3.6.1   The instructor IS able to pause the system at any time during the training simulation for the purpose of administering instruction or procedural recommendations.

B.3.6.2   If a training session begins with the "aircraft in the air" and ready for the performance of a particular procedural task, the instructor IS able to manipulate the following system parameters independently of the simulation:

- Aircraft geographic location,
- Aircraft heading,
- Aircraft airspeed,
- Aircraft altitude, and
- Wind direction, speed, and turbulence.

B.3.6.3   The system IS capable of recording both a horizontal and vertical track of aircraft movement during the entire training session for later playback and review.

B.3.6.4   The instructor IS able to disable any of the instruments prior to or during a training session and IS able to simulate failure of any of the instruments without stopping or freezing the simulation to affect the failure. This includes simulated engine failures and the following aircraft systems failures: alternator or generator, vacuum or pressure pump, pitot static, electronic flight displays, or landing gear or flaps, as appropriate.

B.3.6.5   The ATD HAS a navigational area database that is local (25 nautical miles (NM)) to the training facility to allow reinforcement of procedures learned during actual flight in that area. All navigational data ARE based on procedures as published per 14 CFR part 97. This device uses Navigraph or RealNav Data to support the instrument approach and navigation capabilities.

Advanced ATD Requirements List [Appendix 3 items]

All configurations, as noted in AC 61-136, Appendix C meet these additional AATD design criteria items listed.

The Redbird MX2 meets the following additional AATD CRITERIA:

C.3.1.1   A realistic shrouded (enclosed) or unshrouded (open) cockpit design and instrument panel arrangement representing a specific model aircraft cockpit.

C.3.1.2   Cockpit knobs, system controls, switches, and/or switch panels in realistic sizes and design appropriate to each intended functions, in the proper position and distance from the pilot's seated position, and representative of the category and class of aircraft being represented.

C.3.1.3   Primary flight and navigation instruments appropriately sized and properly arranged that exhibit neither stepping nor excessive transport delay.

C.3.1.4   **Digital Avionics** Panel

C.3.1.5   Global Positioning System **(GPS) navigator** with moving map display.

C.3.1.6   **Two-axis autopilot**, and, as appropriate, a flight director (FD). This is only required when an autopilot is original standard equipment from the aircraft manufacturer.

C.3.1.7   **Pitch trim** (manual or electric pitch trim) IS AVAILABLE permitting indicator movement either electrically or analog in an acceptable trim ratio.

C.3.1.8   An **independent visual system**, panel, or screen that provides realistic cues in both day and night visual flight rules (VFR) and instrument flight rules (IFR) meteorological conditions to enhance a pilot's visual orientation in the vicinity of an airport including:
- Adjustable visibility parameters; and
- Adjustable ceiling parameters.

C.3.1.9   A fixed pilot seat appropriate to the aircraft configuration, including an adjustable height and an adjustable forward and aft seat position.

C.3.1.10   **Rudder pedals** secured to the cockpit floor structure, or that can be physically secured to the floor beneath the device in proper relation to cockpit orientation.

C.3.1.11   **Push-to-talk switch** on the control yoke.

C.3.1.12   A **separate instructor station** PERMITTING effective interaction without interrupting the flight in overseeing the pilot's horizontal and vertical flight profiles in real time and space. This must include the ability to:
1. Oversee tracks along airways, holding entries and patterns, and Localizer (LOC) and glideslope (GS) alignment/deviation (or other approaches with a horizontal and vertical track).
2. Function as air traffic control (ATC) in providing vectors, etc., change in weather conditions, ceilings, visibilities, wind speed and direction, light/moderate/severe turbulence, and icing conditions.
3. Invoke failures in navigation and instruments, radio receivers, landing gear and flaps, engine power (partial and total), and other aircraft systems (pitot, electric, static, etc.) by using either a keyboard or mouse.

The Redbird MX2 model meet the following additional encouraged (not required) AATD CRITERIA:

C.3.2.1   Multi-panel or wrap-around visual system providing a 120 degrees or more of horizontal vision.

C.3.2.2   Automated ATC communications, scenario-based training (SBT), or line-oriented type training in which the instructor can evaluate pilot performance without having to act as ATC.

C.3.2.3   Simulated loss of performance and aerodynamic changes from ice accretion.

C.3.2.4   Realistic aircraft engine sound appropriate to the aircraft configuration, power settings, and speed.

C.3.2.5   A magnetic compass with incremental markings each 5 degrees, that displays the proper lead or lag during turns, and displays incremental markings typical of that shown in the aircraft.

SECTION 5: AIRCRAFT CONFIGURATIONS

List of Previously Approved Configurations:

Beechcraft BE36-S1

- Yoke – Center (See Image 2.1)
- Throttle – Single Engine Complex Lever (T-P-M) (See Image 3.4)
- Traditional analog gauges & 430/530 GPS with KAP140 Autopilot

Beechcraft BE36-G1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Lever (T-P-M) (See Image 3.4)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

Beechcraft BE55-S1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Twin Engine Complex Lever (PP-TT-MM) (See Image 3.6)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Beechcraft BE58-S1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Twin Engine Complex Lever (PP-TT-MM) (See Image 3.6)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Beechcraft BE58-S2

- **Yoke – Center (See Image 2.1)**
- **Throttle – Twin Engine Complex Lever (TT-PP-MM) (See Image 3.5)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Beechcraft BE58-G1
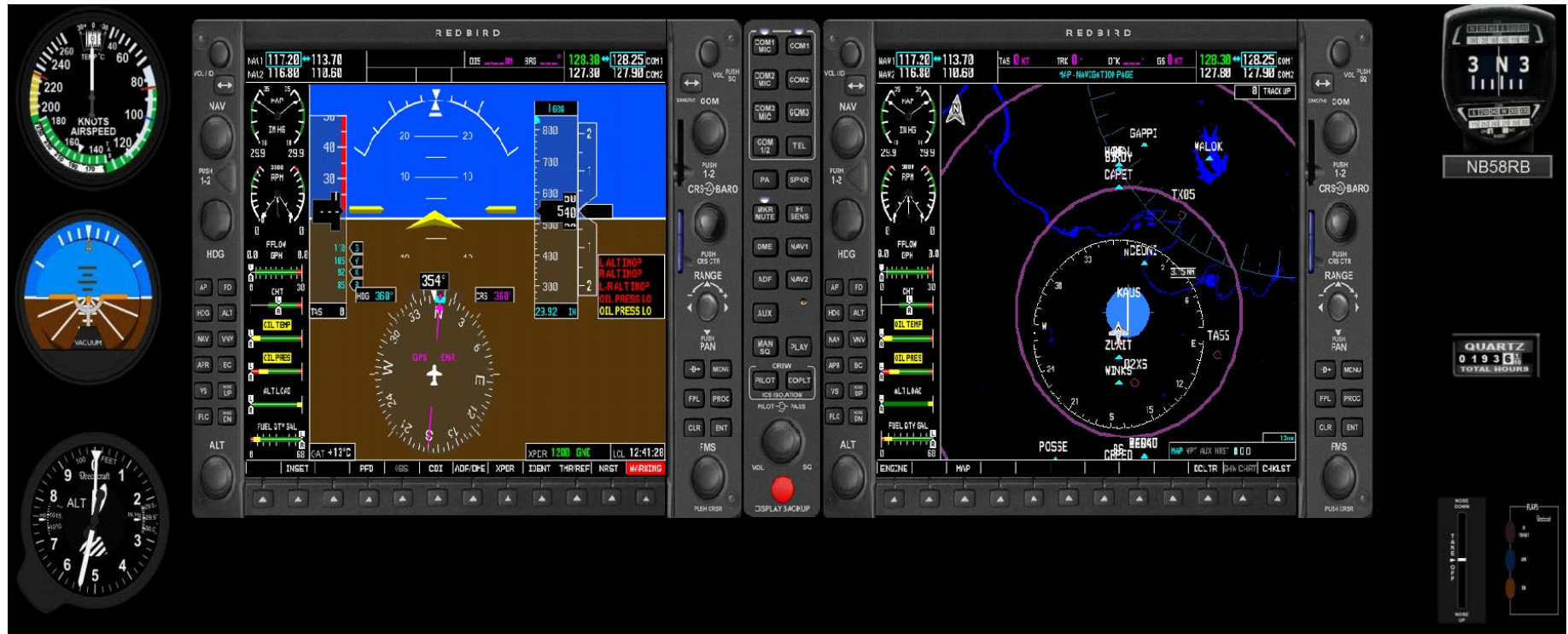
- **Yoke – Center (See Image 2.1)**
- **Throttle – Twin Engine Complex Lever (TT-PP-MM) (See Image 3.5)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

Beechcraft BE76-S1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Twin Engine Complex Lever (TT-PP-MM) (See Image 3.5)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Vernier (T-M) (See Image 3.1)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Cessna C172-S2

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Vernier (T-M) (See Image 3.1)**
- **Traditional analog gauges & KLN94 GPS with KAP140 Autopilot**

Cessna C172-G1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Vernier (T-M) (See Image 3.1)**
- **Glass Cockpit G1000 PFD/MFD & KAP140 Autopilot**

Cessna C172-G2

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Vernier (T-M) (See Image 3.1)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)**
- **Traditional analog gauges & KLN94 GPS with KAP140 Autopilot**

- Yoke – Center (See Image 2.1)
- Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)
- Glass Cockpit G1000 PFD/MFD & KAP140 Autopilot

Cessna C182-G2

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

Cessna C206-S1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Cessna C206-S2

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)**
- **Traditional analog gauges & KLN94 GPS with KAP140 Autopilot**

Cessna C206-G1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)**
- **Glass Cockpit G1000 PFD/MFD & KAP140 Autopilot**

- Yoke – Center (See Image 2.1)
- Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)
- Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot

Cessna C206T-S1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Lever (T-P-M) (See Image 3.4)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Cirrus SR20-G1

- **Yoke – Left (See Image 2.2)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Glass Cockpit Entegra PFD/MFD & Dual 430 GPS with STEC55 Autopilot**

Cirrus SR20-G2

- **Yoke – Left (See Image 2.2)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Glass Cockpit Perspective PFD/MFD & FMS Keypad with Perspective Autopilot**

Cirrus SR22-G1

- **Yoke – Left (See Image 2.2)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Glass Cockpit Entegra PFD/MFD & Dual 430 GPS with STEC55 Autopilot**

Cirrus SR22-G2

- **Yoke – Left (See Image 2.2)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Glass Cockpit Perspective PFD/MFD & FMS Keypad with Perspective Autopilot**

Diamond DA20-S1

- **Yoke – Stick (See Image 2.3)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Traditional analog gauges & 430 GPS with STEC55 Autopilot**

Diamond DA40-G1

- **Yoke – Stick (See Image 2.3)**
- **Throttle – Single Engine Complex Lever (T-P-M) (See Image 3.4)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

Diamond DA40-G2

- **Yoke – Stick (See Image 2.3)**
- **Throttle – Single Engine Complex Lever (T-P-M) (See Image 3.4)**
- **Glass Cockpit G1000 PFD/MFD & KAP140 Autopilot**

Diamond DA42-G1

- **Yoke – Stick (See Image 2.3)**
- **Throttle – Twin Engine Lever (TT-MM) (See Image 3.7)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

Diamond DA42L-G2

- **Yoke – Stick (See Image 2.3)**
- **Throttle – Twin Engine Lever (TT-PP-MM) (See Image 3.5)**
- **Glass Cockpit G1000 PFD/MFD & KAP140 Autopilot**

Liberty LXL2-S1

- **Yoke – Stick (See Image 2.3)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Traditional analog gauges & 430/530 GPS with STEC55 Autopilot**

Liberty LXL2-S2

- **Yoke – Stick (See Image 2.3)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Traditional analog gauges & 430 GPS, EFD1000, SL40 & STEC55 Autopilot**

- **Yoke – Stick (See Image 2.3)**
- **Throttle – Single Engine Complex Lever (T-P-M) (See Image 3.4)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

Piper Warrior PA28-60-S1

- Yoke – Center (See Image 2.1)
- Throttle – Single Engine Lever (T-M) (See Image 3.3)
- Traditional analog gauges & Dual 430 GPS with STEC55 Autopilot

Piper Warrior PA28-60-S2

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Glass Cockpit Entegra PFD/MFD & Dual 430 GPS with STEC55 Autopilot**

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Traditional analog gauges & Dual 430 GPS with STEC55 Autopilot**

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Lever (T-P-M) (See Image 3.4)**
- **Traditional analog gauges & Dual 430 GPS with STEC55 Autopilot**

Piper Arrow PA28-2R-S2

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Lever (T-P-M) (See Image 3.4)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Piper Arrow PA28-2R-G1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Lever (T-P-M) (See Image 3.4)**
- **Glass Cockpit Entegra PFD/MFD & Dual 430 GPS with STEC55 Autopilot**

Piper Arrow PA28-2R-G2

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Lever (T-P-M) (See Image 3.4)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

Piper Seneca PA34-S1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Twin Engine Complex Lever (TT-PP-MM) (See Image 3.5)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Piper Seneca PA34-G1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Twin Engine Complex Lever (TT-PP-MM) (See Image 3.5)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

Piper Seminole PA44-S1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Twin Engine Complex Lever (TT-PP-MM) (See Image 3.5)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Piper Seminole PA44-G1

- Yoke – Center (See Image 2.1)
- Throttle – Twin Engine Complex Lever (TT-PP-MM) (See Image 3.5)
- Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot

Piper Seminole PA44T-S1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Twin Engine Complex Lever (TT-PP-MM) (See Image 3.5)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Socata TB10-S1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

- **Yoke – Stick (See Image 2.3)**
- **Throttle – Twin Engine Lever (TT-CC-PP) (See Image 3.8)**
- **Glass Cockpit G1000 PFD/MFD & STEC55 Autopilot**

Additional configurations included in this version:

Cessna C172RG-S1

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Vernier (T-M) (See Image 3.1)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Cessna C172RG-G2

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Vernier (T-M) (See Image 3.1)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Cessna C182RG-G2

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)**
- **Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot**

Maul MX7-180B

- **Yoke – Center (See Image 2.1)**
- **Throttle – Single Engine Complex Vernier (T-P-M) (See Image 3.2)**
- **Traditional analog gauges & 430/530 GPS with KAP140 Autopilot**

Piper Warrior PA28-60-G2

- Yoke – Center (See Image 2.1)
- Throttle – Single Engine Lever (T-M) (See Image 3.3)
- Glass Cockpit G1000 PFD/MFD & GFC700 Autopilot

Tecnam P2010-G1

- **Yoke – Stick (See Image 2.3)**
- **Throttle – Single Engine Lever (T-M) (See Image 3.3)**
- **Glass Cockpit G1000 PFD/MFD & STEC55 Autopilot**

Performance Table

Single-Engine Performance

Aircraft Model	V$_{SO}$	V$_{S1}$	V$_X$	V$_Y$	V$_A$	V$_{NE}$	V$_G$	V$_{MCA}$	KTAS @ Cruise / 75% power setting	Rate of Climb (fpm) @ (V$_Y$) / Full Power	Single Engine Rate of Climb (V$_{YSE}$)
Beechcraft (A36) Bonanza	53 KIAS	62 KIAS	78 KIAS	96 KIAS	140 KIAS	204 KIAS	110 KIAS	N/A	160 KTAS	1030 fpm	N/A
								5000' -->	166 KTAS	740 fpm	N/A
Cessna 172 Skyhawk	40 KIAS	48 KIAS	56 KIAS	74 KIAS	105 KIAS	163 KIAS	68 KIAS	N/A	116 KTAS	730 fpm	N/A
								5000' -->	119 KTAS	550 fpm	N/A
Cessna 172RG Cutlass RG	39 KIAS	46 KIAS	67* KIAS	84 KIAS	106 KIAS	164 KIAS	73 KIAS	N/A	131 KTAS	800 fpm	N/A
								5000' -->	134 KTAS	590 fpm	N/A
Cessna 182 Skylane	35 KIAS	40 KIAS	58 KIAS	80 KIAS	110 KIAS	175 KIAS	75 KIAS	N/A	129 KTAS	925 fpm	N/A
								5000' -->	135 KTAS	705 fpm	N/A
Cessna 182RG Skylane RG	34 KIAS	39 KIAS	65 KIAS	88 KIAS	112 KIAS	181 KIAS	80 KIAS	N/A	148 KTAS	1140 fpm	N/A
								5000' -->	152 KTAS	778 fpm	N/A
Cessna 206 Stationair	34 KIAS	41 KIAS	65 KIAS	84 KIAS	120 KIAS	183 KIAS	75 KIAS	N/A	141 KTAS	920 fpm	N/A
								5000' -->	145 KTAS	640 fpm	N/A
Cessna 206 Stationair TC	39 KIAS	50 KIAS	74 KIAS	87 KIAS	125 KIAS	182 KIAS	80 KIAS	N/A	139 KTAS	1050 fpm	N/A
								5000' -->	144 KTAS	955 fpm	N/A
Cirrus SR20	56 KIAS	65 KIAS	75 KIAS	85 KIAS	131 KIAS	200 KIAS	96 KIAS	N/A	147 KTAS	828 fpm	N/A
								5000' -->	151 KTAS	566 fpm	N/A
Cirrus SR22	59 KIAS	70 KIAS	78 KIAS	91 KIAS	133 KIAS	201 KIAS	88 KIAS	N/A	170 KTAS	1304 fpm	N/A
								5000' -->	175 KTAS	1015 fpm	N/A
Diamond DA20	37 KIAS	41 KIAS	57 KIAS	65 KIAS	104 KIAS	161 KIAS	72 KIAS	N/A	98 KTAS	675 fpm	N/A
								5000' -->	108 KTAS	440 fpm	N/A
Diamond DA40	49 KIAS	52 KIAS	66 KIAS	66 KIAS	108 KIAS	178 KIAS	73 KIAS	N/A	128 KTAS	1100 fpm	N/A
								5000' -->	136 KTAS	720 fpm	N/A
Liberty XL2	41 KIAS	51 KIAS	70 KIAS	80 KIAS	100 KIAS	157 KIAS	80 KIAS	N/A	NA**	682 fpm	N/A
								5000' -->	NA**	438 fpm	N/A
Mooney M20TN (Type S)	61 KIAS	67 KIAS	84 KIAS	104 KIAS	127 KIAS	194 KIAS	92 KIAS	N/A	182 KTAS	1375 fpm	N/A
								5000' -->	197 KTAS	1275 fpm	N/A
Maul MX7-180B	41 KIAS	53 KIAS	65 KIAS	78 KIAS	109 KIAS	161 KIAS	72 KIAS	N/A	120 KTAS	1000 fpm	N/A
								5000' -->	NA**	NA*	N/A
Piper PA-28-161 Warrior III	44 KIAS	50 KIAS	53 KIAS	79 KIAS	111 KIAS	160 KIAS	73 KIAS	N/A	106 KTAS	644 fpm	N/A
								5000' -->	114 KTAS	396 fpm	N/A
Piper PA-28-181 Archer III	45 KIAS	50 KIAS	60 KIAS	76 KIAS	113 KIAS	154 KIAS	76 KIAS	N/A	121 KTAS	680 fpm	N/A
								5000' -->	125 KTAS	480 fpm	N/A
Piper PA-28R-201 Arrow	55 KIAS	60 KIAS	60 KIAS	90 KIAS	118 KIAS	183 KIAS	79 KIAS	N/A	131 KTAS	840 fpm	N/A
								5000' -->	135 KTAS	580 fpm	N/A
Piper PA-38-112 Tomahawk	46 KIAS	48 KIAS	61 KIAS	70 KIAS	103 KIAS	138 KIAS	70 KIAS	N/A	99 KTAS	710 fpm	N/A
								5000' -->	105 KTAS	460 fpm	N/A
Socata TB10	53 KIAS	60 KIAS	65 KIAS	78 KIAS	122 KIAS	165 KIAS	86 KIAS	N/A	119 KTAS	787 fpm	N/A
								5000' -->	124 KTAS	525 fpm	N/A
Tecnam P2010	50 KIAS	59 KIAS	76 KIAS	82 KIAS	120 KIAS	166 KIAS	85 KIAS	N/A	121 KTAS	713 fpm	N/A
								5000' -->	126 KTAS	510 fpm	N/A

*Gear Up
**Not published by manufacturer

Table 3: Performance (Single-Engine)

Multi-Engine Performance

Aircraft Model	V$_{SO}$	V$_{S1}$	V$_X$	V$_Y$	V$_A$	V$_{NE}$	V$_G$	V$_{MCA}$	KTAS @ Cruise / 75% power setting	Rate of Climb (fpm) @ (V$_Y$) / Full Power	Single Engine Rate of Climb (V$_{YSE}$)
Beechcraft (BE55) Baron	73 KIAS	79 KIAS	91 KIAS	107 KIAS	157 KIAS	224 KIAS	120 KIAS	78 KIAS	168 KTAS	1725 fpm	400 fpm @ 100 KIAS
								5000' -->	180 KTAS	1250 fpm	125 fpm @ 100 KIAS
Beechcraft (BE58) Baron	74 KIAS	84 KIAS	92 KIAS	105 KIAS	156 KIAS	223 KIAS	115 KIAS	84 KIAS	188 KTAS	1725 fpm	395 fpm @ 101 KIAS
								5000' -->	203 KTAS	1325 fpm	150 fpm @ 101 KIAS
Beechcraft (BE76) Duchess	60 KIAS	70 KIAS	71 KIAS	85 KIAS	132 KIAS	194 KIAS	95 KIAS	65 KIAS	148 KTAS	1250 fpm	230 fpm @ 85 KIAS
								5000' -->	160 KTAS	900 fpm	90 fpm @ 85 KIAS
Diamond DA42 L360	57 KIAS	64 KIAS	90 KIAS	90 KIAS	126 KIAS	194 KIAS	NA**	65 KIAS	153 KTAS	1119 fpm	185 fpm @ 85 KIAS
								5000' -->	160 KTAS	1045 fpm	106 fpm @ 85 KIAS
Diamond DA42 NG	62 KIAS	69 KIAS	85 KIAS	90 KIAS	122 KIAS	188 KIAS	NA**	76 KIAS	147 KTAS	1065 fpm	175 fpm @ 85 KIAS
								5000' -->	152 KTAS	996 fpm	122 fpm @ 85 KIAS
Piper PA-34 Seneca	61 KIAS	67 KIAS	76 KIAS	88 KIAS	139 KIAS	204 KIAS	NA**	66 KIAS	157 KTAS	1470 fpm	255 fpm @ 88 KIAS
								5000' -->	171 KTAS	1380 fpm	220 fpm @ 88 KIAS
Piper PA-34T Seneca	62 KIAS	64 KIAS	76 KIAS	92 KIAS	136 KIAS	205 KIAS	NA**	66 KIAS	159 KTAS	1500 fpm	290 fpm @ 92 KIAS
								5000' -->	169 KTAS	1500 fpm	260 fpm @ 92 KIAS
Piper PA-44 Seminole	55 KIAS	57 KIAS	70 KIAS	88 KIAS	135 KIAS	202 KIAS	NA**	56 KIAS	152 KTAS	1350 fpm	225 fpm @ 88 KIAS
								5000' -->	159 KTAS	925 fpm	0 fpm @ 88 KIAS
Piper PA-44T Seminole	56 KIAS	60 KIAS	84 KIAS	88 KIAS	137 KIAS	202 KIAS	NA**	57 KIAS	151 KTAS	1275 fpm	180 fpm @ 88 KIAS
								5000' -->	159 KTAS	1175 fpm	150 fpm @ 88 KIAS
Tecnam P2006	53 KIAS	66 KIAS	80 KIAS	80 KIAS	118 KIAS	167 KIAS	NA**	62 KIAS	142 KTAS	1150 fpm	230 fpm @ 80 KIAS
								5000' -->	139 KTAS	840 fpm	60 fpm @ 80 KIAS

*Gear Up
**Not published by manufacturer

Table 4: Performance (Multi-Engine)

SECTION 6: VISUAL SYSTEM WITH IFR, VFR, DAY, AND NIGHT CAPABILITY

Redbird MX2 Visual System

The visual system is capable of providing a field-of-view of a minimum of 200 degrees horizontally and 24 degrees vertically, for the pilot, including adjustable cloud base and visibility in night, dusk and day scenes.

- The Redbird MX2 provides a means of recording the visual response time for the visual system that is installed.
- The Redbird MX2 visual system is free of optical discontinuities and artifacts that create non-realistic cues.
- The visual system is directly displayed on five (5) LCD monitors inside the cockpit enclosure, situated in an arc around the Pilot. Each monitor is 28 cm tall, and 47 cm wide. Based upon the designated Pilot Eye Point, these monitors provide a horizontal FOV of over 220 (5 monitors) degrees and a vertical FOV of 24 degrees.

Daylight: The visual system provides full color presentations and sufficient surfaces with appropriate textural cues to conduct a visual approach, landing and airport movement. Surface shading effects are consistent with the simulated sun position.

Twilight: The visual system provides full color presentations of reduced ambient intensity, sufficient surfaces with appropriate textural cues that include self-illuminated objects such as road networks, ramp lighting and airport signage, to conduct a visual approach, landing and airport movement. Scenes include a definable horizon and typical terrain characteristics such as fields, roads and bodies of water and surfaces illuminated by representative ownship lighting.

Night: The visual system provides the same as above except the portrayal of reduced ambient intensity; therefore, there is no ground cues that are not self-illuminating or illuminated by ownship lights.

Designated Eye Point: The designated Pilot Eye Point is located 67 cm from the center of the forward most external view monitor, 50 cm from the left most external view monitor and 105 cm from the floor of the simulator enclosure. This point is roughly centered over the pilot's seat when it is adjusted to the middle position, at a height consistent with the height of the pilot's head.
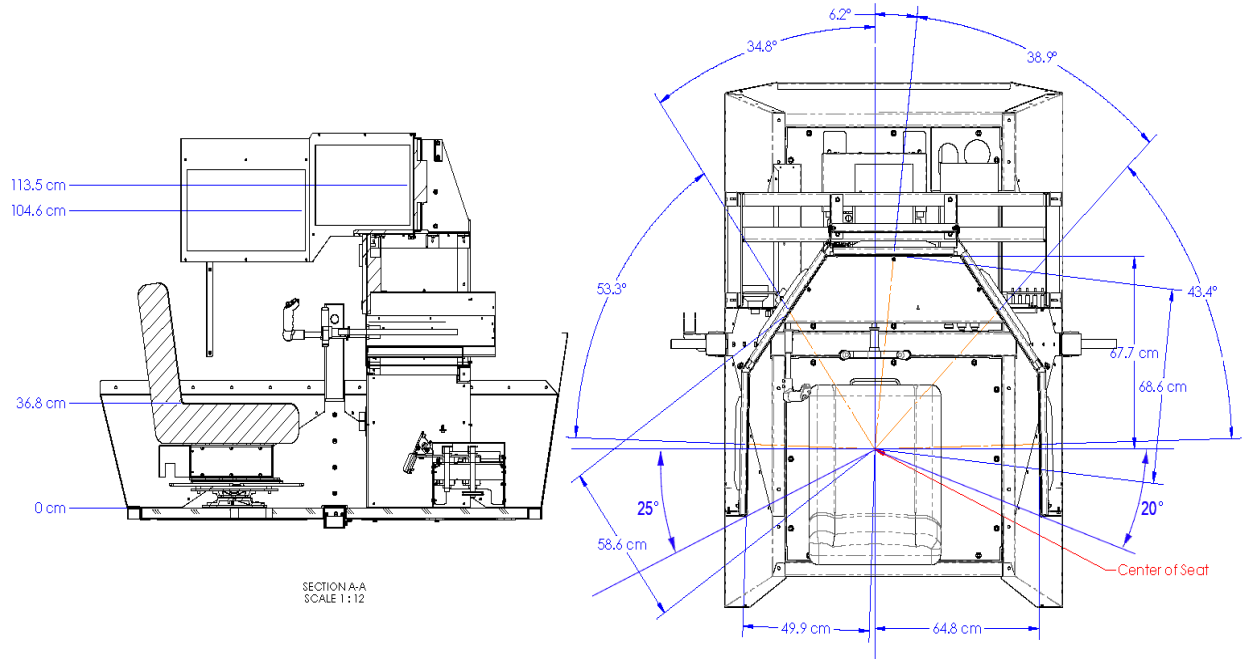
Image 1: Designated Eye Point Diagram

Image 5.4: Visual Displays

SECTION 7: ATD FUNCTIONS & MANEUVERS CHECKLISTS

AIRPLANE ATD FUNCTION VERIFICATION CHECKLIST

Functions and Maneuvers	Yes, No, or N/A
a. Pre-Takeoff	
(1) Engine start	Yes
(2) Taxi and brake operation	Yes
b. Takeoff	
(1) Run-up and powerplant checks	Yes
(2) Acceleration characteristics	Yes
(3) Nose wheel and rudder steering	Yes
(4) Effect of crosswind	Yes
(5) Instrument	Yes
(6) Flap operation	Yes
(7) Landing gear operation (if retractable)	Yes
c. In-Flight Operations	
(1) **Climb**	
(i) Normal and max. performance	Yes
(ii) One engine inoperative procedures (Multiengine only)	Yes
(2) **Cruise**	
(i) Correct performance characteristics (speed vs. power)	Yes
(ii) Normal and steep turns	Yes
(iii) Approach to stalls, (i.e. stall warning), stalls. Execute from takeoff, cruise, and approach and landing configurations.	Yes
(vi) In flight engine shutdown (multi-engine only)	Yes
(v) In flight engine start (multi-engine only)	Yes
(vi) Fuel selector function	Yes
(3) **Approach**	
(i) Normal (with & without flaps) Check gear horn warning if applicable	Yes
(ii) Single engine approach and landing (multi-engine)	Yes
(iii) Best glide no power	Yes
(iv) Landings	Yes
d. Instrument Approaches	
(1) **Nonprecision**	
(i) GPS and LPV	Yes
(ii) GPS - WAAS (optional)	Varies by configuration
(iii) All engines operating	Yes
(iv) One engine inoperative (Multi-engine only)	Yes
(v) Approach procedures (VOR, VOR/DME, LOC procedures on an ILS, LDA, RNAV (RNP) or RNAV (GPS) to LNAV, LNAV/VNAV or LPV)	Yes
(2) **Precision**	
(i) ILS	Yes
(ii) GLS (optional)	No
(iii) Effects of Crosswind	Yes
(iv) One Engine Inoperative (Multi-engine only)	Yes
(v) Missed Approach	Yes
(A) Normal	Yes

(B) With One Engine inoperative (Multi-engine only)	Yes
e. Surface Operations (Post Landing)	
(1) Approach and landing roll	Yes
(2) Braking operation	Yes
(3) Reverse thrust operation, if applicable	Yes
f. Any Flight Phase	
(1) **Airplane and Power Plant Systems**	
(i) Electrical, mechanical, or hydraulic	Yes
(ii) Flaps	Yes
(iii) Fuel selector and oil temp/pressure	Yes
(vi) Landing gear (if applicable)	Yes
(2) **Flight Management and Guidance Systems**	
(i) Two axis auto pilot (if standard equipment)	Yes
(ii) Flight director (AATD only) and system displays (if installed)	Yes
(iii) Navigation systems and optional display configurations	Yes
(iv) Stall warning systems avoidance	Yes
(v) Multi-function displays (PFD/MFD) if applicable	Yes
(3) **Airborne Procedures**	
(i) Holding	Yes
(ii) Uncoordinated turns – slipping and skidding demo	Yes
(iii) Configuration and power changes and resulting pitch changes	Yes
(iv) Compass turns and appropriate errors (if installed)	Yes
(4) **Simulated Turbulence in Flight** (light, moderate, severe)	Yes
(4) **Parking and Engine Shutdown**	
(i) Systems operation	Yes
(ii) Parking brake operation (if installed)	Yes
g. Can simulate engine failure, including failures due to simulated loss of oil pressure or fuel starvation.	(Except oil pressure loss)
h. Can simulate the following equipment or system failures:	
(1) Alternator or generator failure.	Yes
(2) Vacuum pump/pressure failure and associated flight instrument failures.	Yes
(3) Gyroscopic flight instrument failures.	Yes
(4) Pitot/static system malfunction and associated flight instrument failures.	Yes
(5) Electronic flight deck display malfunctions.	Yes
(6) Landing gear (if retractable) or flap malfunctions	Yes
i. Independent Instructor Station Requirements (AATD only)	
(1) Displays published airways and holding patterns.	Yes
(2) Displays airplane position and track.	Yes
(3) Displays airplane altitude and speed.	Yes
(4) Displays NAVAIDs and airports.	Yes
(5) Can record and replay airplane ground track history for entire training session.	Yes
(6) Can invoke instrument or equipment failures.	Yes

During the initial start of the trainer, the computer component "self-check" program verifies that all the features of the trainer are in working order. It is not possible to continue the training session unless the problem is resolved, and all the components are functioning properly.

During the initial start-up the ATD has the following **Screen Statement** is displayed on the instructor station or visual display before the trainer is used for training.

"All the flight instruments required for visual and instrument flight rules listed in part 91.205 must be functional at the start of the simulated flight session. Temporary instrument or equipment failures are permitted when practicing emergency procedures. If this simulated flight session will be used for instrument experience or currency requirements, the visual component must be configured to Instrument Meteorological Conditions [IMC] during the simulated flight session, including execution of instrument approaches from the final approach fix until reaching Decision Height [DH], Decision Altitude [DA], or Minimum Decent Altitude [MDA] as appropriate."

Notice: Any changes or modifications to this training device that have not been reviewed, evaluated, and approved in writing by General Aviation and Commercial Division will terminate FAA approval.